			GBGS	SCHEME			
USN						15AE63	
Sixth Semester B.E. Degree Examination, Aug./Sept.2020							
Aircraft Performance							
Tin	no: ?	hre			Max M	Max Marks: 80	
						aiks. 00	
Note: Answer any FIVE full questions, choosing ONE full question from each module.							
1	a. b.	Calculate and plot 73000 lb. Air data @30,000 ft $\rho_{\infty} = 8$ Derive the equation	the thrust required a : S = 950 ft <sup>2</sup> , AR .9068×10 <sup>-4</sup> slug/ft <sup>3</sup> . n which shows the	<b><u>Iodule-1</u></b> curve at an altitude $L = 5.92$ , $C_{DO} = 0$ variation of drag with	e of 30,000 ft, assuming 0.015 and K = 0.08, $V_{\alpha}$ ith altitude, velocity and	a weight of = 500 fl/s, (10 Marks) weight. (06 Marks)	
OR							
2	a. Derive the equation which shows, $V_{\infty}$ for a given $T_R$ depends on $T_R/W$ , W/S, $C_{DO}$ and K						
	b.	Describe the effec	t of altitude on the p	oint corresponding	to the minimum thrust r	equired. (07 Marks)	
		2	<u>N</u>	Iodule-2			
3	a.	Show that $(R/C)_{ma}$	x, being dominated	by the thrust o we	ight ratio, decrease with	an increase	
	b.	Derive the equation	n which shows that	$(V_V)_{min}$ occurs at (	$C_L^{3/2}/C_D$ )max	(06 Marks)	
-				OR			
4	a. b.	Derive the general Describe Hodogra	equation for range. ph for unpowered fl	ight.		(08 Marks) (08 Marks)	
				Iodule-3			
5	a.	Derive the equation	n to show	<u>iouure c</u>			
		$V_{\left(C_{L}^{3/2}/C_{D}\right)_{max}}$	$< V_{(C_L / C_D)_{max}} < V_{(C_L^{-1})}$	$^{2}$ / C <sub>D</sub> ) <sub>max</sub>	<u>,</u>	(10 Marks)	
	b.	Derive the express	ion for variation of	(L/D) with velocity	4.	(06 Marks)	
				OR			
6	a. b.	Derive Range for J Derive Endurance	for propeller-driven air	plane and jet airplai airplanes.	ne.	(10 Marks) (06 Marks)	
						、	
7	a. b.	Explain Intermedi Derive the expres performance.	<u>N</u> ate segments of the ssion for ground ro	Ground Roll with r	neat diagram. n parameters that gove	(07 Marks) rn take off (09 Marks)	
				OR			
8	a. b.	Describe the landi Derive S <sub>g</sub> that illus	ng path and landing strates the design pa	distance with neat rameter that govern	diagram. n landing performance.	(07 Marks) (09 Marks)	
				1 of 2			

•

•

## 15AE63

## Module-5

9 a. Derive the expression for turn radius and turn rate. b. Derive the expression for minimum turning radius.

(06 Marks) (10 Marks)

## OR

- a. Derive the expression for pull up and pull down maneuvers. (06 Marks) 10 b. Derive the equation that gives the load factor for a given velocity and thrust to weight ratio.
  - c. Describe Ground effects.

(06 Marks) (04 Marks)